Experimental evaluation in Heat Transfer Augmentation using hexagonal finned surfaces

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ABSTRACT

An attempt has been made for the experimental evaluation to study the effect of the hexagonal finned surface on the local heat transfer coefficients between the impinging circular jet and flat plate. Reynolds number is varied between 7000 and 30000 based on the nozzle exit condition and jet to plate spacing between 0.5 to 6 times of the nozzle diameter. The fins used are in the form of hexagonal prism of side 2.04 mm and height of 2 mm spaced at a pitch of 7.5 mm on the target plate. It is observed that there is increase in the heat transfer coefficient up to 68 % depending on the nozzle plate spacing and the Reynolds number.

Keywords: Impinging air jet, Heat transfer augmentation, Finned surface, Nusselt number.

INTRODUCTION

Impinging jets have received considerable attention due to their inherent characteristics of high rates of heat transfer besides having simple geometry. Various industrial processes involving high heat transfer rates apply impinging jets. Heat transfer rates in case of impinging jets are affected by various parameters like Reynolds number, nozzle plate spacing, radial distance from stagnation point, Prandtl number, target plate inclination, confinement of the jet, nozzle geometry, curvature of target plate, roughness of the target plate, low scale turbulence intensity i.e., turbulence intensity at the nozzle exit. Gardon and Cobonpue$^1$ have reported the heat transfer distribution between circular jet and flat plate. They also studied effect of multiple two-dimensional jets on the heat transfer distribution$^3$. Baughn and Shimizu$^4$ and Hrycak$^5$ have conducted experiments of heat transfer between round jet and flat plate employing different methods of surface temperature measurement.

Review of the experimental work on impinging jets is done by Martine$^7$, Jambunathan et. al.$^8$ and Viskanta$^9$. Hansen and Web$^{10}$ have studied the effect of the modified surface on the heat transfer between impinging circular nozzle and the flat plate. However their data reflects the average Nusselt number variation rather than local data because of the large thickness of the target plate used.

Literature review suggests that there is lack of information on local heat transfer data in case of the finned surfaces. Therefore, the aim of the present work is to study the effect of the hexagonal finned surface on the local heat transfer coefficients between the impinging circular jet and flat plate. The experimental parameters are the Reynolds number varied between 7000 and 30000 based on the
nozzle exit condition and jet to plate spacing of 0.5, 1, 2, 4, 6 times the nozzle diameter.

**EXPERIMENTAL**

The experimental set up layout is depicted in Fig. 1 (a). Air jet is supplied by a three-cylinder two-stage air compressor through a calibrated orifice flow meter. Air filter and Pressure regulator are installed upstream of the orifice flow meter to filter the air and to maintain the downstream pressure at a value of $4 \pm 0.05$ bar. The flow rate is controlled by two needle valves, one on each side of the orifice flow meter. The nozzle, which directs the air until it impinges upon a heated target plate, is constructed with a 7.35 mm inner diameter aluminum pipe of length to diameter ratio of 83, which is almost same as that of Lyttle and Webb. This length is sufficient to ensure fully developed flow over the Reynolds number range investigated. The end of the nozzle pipe is machined perpendicular to the nozzle axis. The impinging plate is constructed using 1mm thick stainless steel plate of size 120mm x 120 mm. Nichrome heater of size 100mm x 100mm is packed between impinging plate and a bakelite support plate with mica sheets in between to isolate the impinging plate from the heater electrically. To ensure that the impinging plate makes perfect contact with the heater, the impinging plate and bakelite support are tightened together with nuts and screws as shown in Fig. 1 (b). This assembly is then insulated from all sides except

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**Fig. 1: Experimental Test Set-Up**

1. Air filter  
2. Pressure regulator  
3. Needle valve  
4. Orifice flow meter  
5. Water manometer  
6. Computer  
7. Reciprocating compressor  
8. Target plate assembly  
9. Thermoteknix infrared camera  
10. 2D Traverse system  
11. Nozzle pipe  
12. Target plate  
13. Leads for heater  
14. Bakelite support  
15. Mica sheet  
16. Heater  
17. Screw and nut assembly
Fig. - 2: Comparison of the Nusselt number for Smooth and Hexagonal finned surface at different $Z/D$'s and Reynolds number of 23000.
Fig. 3: Comparison of the Nusselt number for smooth and hexagonal finned surface at a $Z/D$ of 6 for different Reynolds number

the impinging surface by ceramic wool and thermocole to reduce the back and side conduction losses. A two-dimensional traverse system is used to locate the target plate at a given position with respect to nozzle exit.

Electric power is supplied to the heater through variac. The voltage and the current are measured by digital panel meters. A K-type Chromel-Alumel thermocouple junction is soldered on the target plate at its extreme end to know about the steady state. The output of the thermocouple is measured by ‘Agronic’ milivoltmeter. Similarly, thermocouples are attached on cooled surface along the flow direction from the stagnation point with an interval of 5 mm between each thermocouple to measure wall temperature distribution.

RESULTS AND DISCUSSION

The heat transfer data is collected for five different nozzle plate spacing of 0.5, 1, 2, 4 and 6 times the diameter of the nozzle at a given Reynolds number of 23000 as shown in figure 2(a) to 2(e) and for six Reynolds numbers of 7000, 12500, 16000, 21500, 25000 and 30000 at given nozzle plate spacing of 6 times the nozzle diameter for smooth and finned surfaces as shown in figure 3(a) to 3(f). The heat transfer data for hexagonal finned surface shows an increase in Nusselt number as compared to smooth plate on the base surface and reduction in the Nusselt number on the top of the fins as compared to smooth surface. The total increase in the area of the hexagonal finned surface is about 41% as compared to the smooth surface. The increase in heat transfer is higher for \( Z/D \) of 1, and 2 as compared to other nozzle plate spacings. For \( Z/D \) of 1 the increase in the heat transfer on the base surface ranges between 24% to 49% depending on the radial location from the stagnation point and decrease in the heat transfer on the top of the fin at the stagnation point is about 50% as shown in fig.2(b).

Figure 3(a) to 3(f) shows the effect of Reynolds number on the heat transfer distribution at a nozzle – plate spacing of 6\( D \) and observed that there is increase in the heat transfer on hexagonal finned surface as compared to smooth surface.

Conclusions

An experimental evaluation is conducted in the present work to study the effect of the finned surface on the local heat transfer coefficients between the impinging circular jet and flat plate. The experimental parameters are the Reynolds number varied between 7000 and 30000 based on the nozzle exit condition and jet to plate spacing of 0.5, 1, 2, 4, 6 times the nozzle diameter. The configuration of surface in the form of fins studied, are hexagonal fins of side 2.04 mm and height of 2 mm spaced at a pitch of 7.5 mm on the target plate

The hexagonal finned surface increases the heat transfer coefficient from the target plate, of about 22 to 68% at the base surface depending on the nozzle plate spacing and the Reynolds number in their ranges studied. The decrease in the Nusselt number at the top of the fin is about 41 to 70%. The increase in the heat transfer for finned surface may be because of the increase in the swirl i.e. superposition of tangential velocity component onto the axial flow, which affects the turbulence characteristics of the flow and the decrease in the heat transfer coefficient at the top of the fins is because of the increase in the heat transfer area.

Nomenclature

\( D \) Diameter of the nozzle exit (m)
\( h \) Heat transfer coefficient (W/m².K)
\( Nu \) Nusselt number (Dimensionless)
\( r \) Radial distance from the stagnation point (m)
\( Re \) Reynolds number based on nozzle exit condition (Dimensionless)
\( Z \) Nozzle plate spacing (m)
REFERENCES


